Comparative Study on Different Turbulent Models for Internal Flow in a Three Dimensional Car Model Using CFD

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ABSTRACT: Human comfort is one of the major concerns in designing any vehicle passenger cabin. Investigation on air flow requirement and temperature distribution becomes very much essential in order to ensure thermal comfort and hence considering this as top priority, an air conditioning system need to be developed without affecting the vehicle performance. In present study, a symmetric three dimensional car model is considered and passengers are with no heat generation. Numerical investigation is carried out using ANSYS Fluent to understand the velocity and temperature distribution in passenger car cabin for two different inlet velocity conditions. Also a comparative analysis is been carried out for the same using different turbulent models such as standard k-epsilon model, RNG k-epsilon model and SST k-omega model to understand computation time and results variation. From the analysis, temperature results obtained around the passengers found to be in thermal comfort zone as per requirement of ASHRAE 2003 and among three turbulent models, SST k-omega model gives better results with same accuracy in less computation time.

KEYWORDS: Thermal comfort, ANSYS Fluent, Turbulent models

I. INTRODUCTION

Travelling is the major part in Humans life during in where would feel relaxed and need to be memorable. Vehicle used for travelling should be with comfort and safety hence an enclosed cabin is required with an HVAC system. An enclosed air conditioned cabin results in Thermal or human comfort, Smoke and dust avoid travel, protection from weather condition and other environmental influences. In designing such a cabin it is extremely important to ensure that HVAC system incorporated should not affect the vehicle performance and hence investigation on airflow requirement and temperature distribution is essential to ensure vehicle performance optimization.

Computational Fluid Dynamics (CFD) tool is being playing a major role in designing such a complex systems involving huge and complex flow and heat equations which is impossible through human analytical evaluation and calculation. Hence CFD tool has become a major role player in Automobile industry in aerodynamic design vehicle, Engine optimization combustion and thermal comfort cabins.

In present paper, CFD analysis using different turbulent models is been carried out to investigate the air flow requirement and temperature distribution for different velocity conditions using commercial CFD tool ANSYS Fluent.

II. LITERATURE SURVEY

A literature survey is been conducted to find if any analysis on an internal flow of a car has been conducted on similar ground of present work. Several internal flow analysis conducted by few individuals were found but those analysis was dependent on other factors and parameters. The summary of those reports are as given below.

[1] Haslinda Mohamed Kamar et.al., 2012

• This paper presents a numerical study on the temperature field inside a passengers compartment of a “Proton Wira” car using CFD
• Aim of the study was to investigate air flow requirement and air temperature in passenger compartment considering glazing effect in analysis
• The passengers are considered to be solid bodies without heat generation
• Results showed that, when glazing material is used the air temperature around the driver was slightly lower than at rear passenger and when Green insulated laminate is used the air temperature in both front and rear end was found to be reduced

• This research article describes numerical adiabatic flow simulation conducted for “mute- electric car”.
• The car has seven vents. Three are under the windshield window, two on dashboards sides and two in the passengers foot well in which air enters the cabin through these vents
• Heat generation by the passenger was not considered
• Analysis was carried out only for air velocities in a cabin
• Air velocities were compared for with and without passengers using different turbulent models such as standard k-epsilon, RNG k-epsilon and SST k-omega model in comparison with three different wall treatments such as standard, scalable and enhanced wall treatments and compared with experimental results
• Study finds SST k-omega model gives better overall performance compared to standard k-epsilon and RNG k-epsilon models

• Paper presents the study on air flow inside a passenger car cabin with varied angle of inlet vents
• Analysis were carried out in three stages; In first stage, temperature was measured experimentally. In second stage 3D cad model was generated by measuring design data and finally CFD car model was developed to validate the experimental data
• Results shown that faster cooling and uniform air distribution is possible through varied angle of vents
• Paper highlights the empirical formulas for calculating solar heat flux and convective heat transfer co-efficient and also air velocities to be chosen for analysis through experimental data’s

It is observed that many literatures are available on similar grounds of present work. However only few important articles are discussed under this section, remaining journals related is listed in reference. These articles help as a ready reckoner in choosing turbulent models, boundary conditions and discretization schemes.

III. METHODOLOGY

In any analysis process, the steps involved are creating Geometry, Meshing and Analysis. The present work is carried in the following process as shown in Figure 1.

Figure 1: Analysis process
The three dimension model of car geometry is made by using CATIA, Meshing and Analysis is carried out using ANSYS products.

3.1 Geometry

The three dimension car model is a “Proton Wira” consists of four passengers, front end with inlet and outlet vents, rear end, side wall and a roof wall. For simplicity, model is considered to be planar symmetry as shown in Figure 2. The front end of the cabin consists of two inlet vents one is side vent and other is centre vent and bottom of front end consist of outlet vent. The passenger body is made flat for simplicity and placed in 90 degree view.

![Figure 2: Three dimension car model](image)

3.2 Meshing

The three dimension mesh model is as shown in Figure 3 and side view of the same model is as shown in Figure 4. The Mesh is tetrahedral type with 353237 elements and 69039 nodes. The same mesh model is used for all the different cases of analysis.

![Figure 3: Three dimension mesh model](image)  ![Figure 4: Side view of mesh model](image)

The analysis is been carried out for above mesh model for two different inlet velocity conditions as shown in Table 1, in comparison with three different turbulent models such as standard k-epsilon model, RNG k-epsilon model and SST k-omega model.
Table 1: Different inlet velocity conditions

<table>
<thead>
<tr>
<th>Location</th>
<th>Case 1</th>
<th>Case 2</th>
</tr>
</thead>
<tbody>
<tr>
<td>Side inlet vent</td>
<td>2</td>
<td>3</td>
</tr>
<tr>
<td>Centre inlet vent</td>
<td>2</td>
<td>3</td>
</tr>
</tbody>
</table>

For all the cases, Residuals for continuity, x-y-z velocity and turbulent models are $10^{-3}$ and for energy $10^{-6}$ is considered for convergence.

The convergence criterion for continuity, x-y-z velocity, energy and standard k-epsilon (ske) for case 1 is as shown in Figure 5 and for case 2 is as shown in Figure 6.

The convergence criterion for continuity, x-y-z velocity, energy and RNG k-epsilon (rngke) for case 1 is as shown in Figure 7 and for case 2 is as shown in Figure 8.

The convergence criterion for continuity, x-y-z velocity, energy and SST k-omega (sstkw) for case 1 is as shown in Figure 9 and for case 2 is as shown in Figure 10.
As the summary, Table 2 shows the comparison on convergence criteria or iterations for different turbulent models.

<table>
<thead>
<tr>
<th>Turbulent Models</th>
<th>Inlet velocity</th>
<th>Iterations</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>condition in m/s</td>
<td></td>
</tr>
<tr>
<td>ske</td>
<td>case 1</td>
<td>303</td>
</tr>
<tr>
<td></td>
<td>case 2</td>
<td>407</td>
</tr>
<tr>
<td>rngke</td>
<td>case 1</td>
<td>337</td>
</tr>
<tr>
<td></td>
<td>case 2</td>
<td>461</td>
</tr>
<tr>
<td>sstkw</td>
<td>case 1</td>
<td>256</td>
</tr>
<tr>
<td></td>
<td>case 2</td>
<td>263</td>
</tr>
</tbody>
</table>

From Table 2, it is observed that the mesh with sstkw model converged in less computation time or iterations.

3.3 Solution Methodology

Present investigation is been carried out based on the following assumptions:
- Flow is turbulent
- Flow is incompressible
- Flow is at steady state condition
- Passengers are with no heat generation
- Walls are stationary and with no slip shear condition

The other boundary conditions incorporated for the mesh model is as shown in Table 3.

<table>
<thead>
<tr>
<th>Location</th>
<th>Boundary condition type</th>
<th>Case 1</th>
<th>Case 2</th>
</tr>
</thead>
<tbody>
<tr>
<td>Side inlet vent</td>
<td>Velocity [1]</td>
<td>2</td>
<td>3</td>
</tr>
<tr>
<td>Centre inlet vent</td>
<td>Velocity [1]</td>
<td>2</td>
<td>3</td>
</tr>
<tr>
<td>Outlet vent</td>
<td>Pressure</td>
<td>0 (gauge pressure)</td>
<td>0 (gauge pressure)</td>
</tr>
<tr>
<td>Roof wall</td>
<td>No slip and Heat flux [1]</td>
<td>260 W/m²</td>
<td>260 W/m²</td>
</tr>
<tr>
<td>Side wall</td>
<td>No slip</td>
<td>-</td>
<td>-</td>
</tr>
</tbody>
</table>

Boundary conditions are maintained same for all the different cases of analysis.
The flow is solved for three different turbulent models such as standard k-epsilon, RNG k-epsilon and SST k-omega. SIMPLE discretization scheme is used to solve transport momentum equation and turbulence parameters.

### 3.4 Mathematical models

Each and every cell in the computational domain is converged for basic governing equations such as Mass conservation equation, Momentum conservation equation and Energy conservation equation. These basic equations are valid for both compressible and incompressible flows. The equations with no time derivative is said to be steady state flow equations.

Since flow is turbulent additional transport equations are solved as described below:

**Standard k-epsilon (k-ԑ) model:**
It is a semi-empirical model and is also known as two equation model in which two separate transport equations for turbulent kinetic energy (k) and its dissipation rate (ԑ) used independently to determine the turbulent velocity and length scales.

The model transport equations for ‘k’ is derived from exact equation, while the model transport equations for ‘ԑ’ was obtained using physical reasoning.

It is one of the popular models in industrial flow and heat transfer analysis due to its Robustness, economy and reasonable accuracy for a wide range of turbulent flows.

**RNG k-epsilon (k-ԑ) model:**
Mathematical technique called “Renormalization group” (RNG) is used for Navier-Stokes equation to derive RNG based k- Epsilon (k-ԑ).

Though it seems similar to standard k- Epsilon (k-ԑ) model but few refinements are included as below:

- It has additional term in ԑ equation, which improves the accuracy for rapidly strained flows
- It enhances accuracy for swirling flows
- It accounts for low Reynolds number effects

These features make the RNG k- Epsilon (k-ԑ) model more accurate and reliable than Standard k-epsilon (k-ԑ) model.

**Shear-Stress Transport (SST) k-ԑ model:**
It is an empirical model based on model transport equations for turbulent kinetic energy (k) and specific dissipation rate (ԑ).

This model predicts accurate measurements for the following:

- far wakes
- mixing layers
- plane, round and radial jets

Thus this model is applicable to wall bounded flows and free shear flows.

All above mathematical models cannot be solved analytically and hence FLUENT based solver is used to solve the complex flow equations.

For further study on these models refer FLUENT theory guide [4].

### 3.5 Results and discussion

As stated in the earlier sections, the analysis is carried out for two cases of inlet velocity conditions comparing with different turbulent models such as Standard k-epsilon (ske), RNG k-epsilon (rngke) and SST k-omega (sstkw). The results obtained for those are discussed below:

The plane selected for result visualization is as shown in Figure 11.
3.5.1 Velocity distribution

3.5.1.1 Case 1: For side and center inlet velocity condition of 2 m/s

Figure 12: Velocity contour for ske-case 1

Figure 13: Velocity contour for rngke-case 1

Figure 14: Velocity contour for sstk-w-case 1

3.5.1.2 Case 2: For side and center inlet velocity condition of 3 m/s

Figure 15: Velocity contour for ske-case 2

Figure 16: Velocity contour for rngke-case 2
Figure 12, Figure 13 and Figure 14 shows the velocity contours for ske, rngke and sstkw for case 1 and Figure 15, Figure 16 and Figure 17 shows the velocity contours for ske, rngke and sstkw for case 2. Comparison with three turbulent models for the case-1 and case-2, the maximum velocity of approximately 3m/s and 4.5m/s found at outlet. Also it can be observed that the velocity distribution for sstkw is better than ske and rngke though there is negligible variation in the results.

3.5.2 Temperature distribution

3.5.2.1 Case 1: For side and center inlet velocity condition of 2 m/s

- Figure 18: Temperature contour for ske- case 1
- Figure 19: Temperature contour for rngke- case 1
- Figure 20: Temperature contour for sstkw- case 1

3.5.2.2 Case 2: For side and center inlet velocity condition of 3 m/s

- Figure 21: Temperature contour for ske- case 2
- Figure 22: Temperature contour for rngke- case 2
Figure 18, Figure 19 and Figure 20 shows the temperature contours for ske, rngke and sstkw for case 1 and Figure 21, Figure 22 and Figure 23 shows the temperature contours for ske, rngke and sstkw for case 2. Comparison with three turbulent models for both case-1 and case-2, the maximum temperature of approximately 33°C found at front and rear ends sharp edges (dead zone) as because there is no air flow at those zones. Also found that there is negligible increase in temperature in the domain for both the cases. The temperature at the passengers section is found to be 25 – 28°C. According to the requirements for thermal comfort in summer conditions interior air temperature should be in the range of 23 to 28°C (ASHRAE 2003) [5].

3.5.3 Velocity streamlines

3.5.3.1 Case 1: For side and center inlet velocity condition of 2 m/s
3.5.3.2 Case 2: For side and center inlet velocity condition of 3 m/s

Figure 27: Velocity streamlines for ske- case 2

Figure 28: Velocity streamlines for rngke- case 2

Figure 29: Velocity streamlines for sstkw- case 2

Figure 24, Figure 25 and Figure 26 shows the velocity streamlines for ske, rngke and sstkw for case 1 and Figure 27, Figure 28 and Figure 29 shows the velocity streamlines for ske, rngke and sstkw for case 2. Streamlines signifies the circulation zones and air distribution in the domain.

It can be observed that sstkw model gives better velocity streamlines compared to ske and rngke for both the cases.

As summary, Figure 30 and Figure 31 shows the graph plotted for the results of maximum velocity and maximum temperature for different turbulent models. As discussed earlier and from the graph the variation in the results is negligible.

Figure 30: Graph of maximum velocity vs. turbulent models

Figure 31: Graph of maximum temperature vs. turbulent models
IV. CONCLUSIONS

The analysis is been carried out on three dimension car cabin for two different inlet velocity conditions in comparison with three different turbulent models such as standard k-epsilon, RNG k-epsilon and SST K-omega, from the results obtained the following are the conclusions drawn:

1. CFD tool is a powerful solver for complex internal flow analysis in determining the velocity and temperature distribution
2. CFD provides the insight of the problem through color post scripts of results. This help the designers for decision making
3. Results obtained from SST K-omega model is with less computation time and iterations with same accuracy compared to k-epsilon and RNG k-epsilon models
4. Velocity distribution contours, temperature distribution contours and velocity streamline plots obtained for SST K-omega model is better than compared to k-epsilon and RNG k-epsilon models though there are negligible variation in the results
5. Temperatures found at the passenger section is about 25-28°C which is in the acceptable range as per the requirement of ASHRAE 2003

It can also be concluded that the present study is no more exhaustive and can be carried as future scope of work by changing the vent shapes, vent positions, human with heat generation and considering the glazing effects etc. The results obtained from the present study help the designer as a ready reckoner in making decision faster in choosing the Boundary conditions, turbulent models and discretization schemes.

REFERENCES