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## **Experimental Study of Mechanical Behaviour of Fiber Matrix Interfaces of The Epoxy and Polyester Reinforced with En-8 Carbon Steel**

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**ABSTRACT**: An Experimental study of the fiber matrix Interfaces on the stress of the Epoxy andpolyester reinforced with long ferrous (En-8 carbon steel) and Nonferrous (Aluminium) is desired. Tensile test, Impact test Torsion test shows the existences of interfaces between fiber & matrix which are linear and Nonlinear of the material with two different types of long fibers. The number of debonding stress are characterizing each interface of the different debonding length. Mechanical properties are calculated by the manual testing using Tensile, Torsion, and Impact tests.

KEYWORDS: Bimetal, Torsion test, bounding, Matrix composites

#### **I.INTRODUCTION**

A composite material is a material made from two or more constituent significantly different physical or chemical properties that, when combined, produce a material with characteristics different from individual components. The individual components remain separate and distinct within the finished structure. The new material may be preferred for many reasons: common examples include materials which are stronger, lighter, or less expansive when compared to traditional materials.

Typical engineered composite materials include:

- Mortars, concrete
- Reinforced plastics, such as fiber-reinforced polymer
- Metal composites
- Ceramics composites

Many composite materials are composed of just two phase; one is termed the matrix, which continuous and surrounds the other phase, often called the dispersed phase. The properties of composites are a function of the properties of the constituent phases, their relative amounts, and the geometry of the dispersed phase.

#### **II.OBJECTIVES**

Our objective is to:

- Select the right material for design and analysis.
- Make the composite sample workpiece from designed material.
- Performing Tensile, Torsion, and Impact test on finished product and tabulate the results.



#### **IV.MATRIX FORMATION**

We had make the composite matrix of metals combination of two that is Steel vs. polyester and Steel vs. Epoxy. These two composite matrix are processed to bounding of 50%, 100%. The bounded work sample is subjected to Tensile, Torsion and Impact test. Tensile test is done by UTM machine. Torsion is done by Twisting machine, Impact test is accomplished by Izoid and charpy experiment test.



Fig 1. Composite workpiece sample

s.no	Material	Quantity
1		4
	EN-8 steel rod	
2		4
	MS square rod	
3		4
	MS square	
4		8
	Steel pipe	
5		8
	Washer	
6		1/2kg
	Epoxy resins	
7		1/4kg
	Polyester resins	
8		1/4kg
	Accelerator	
9		1/4kg
	Catalyst	
10	Polyvinyl(releasing agent)	1/4kg
11	Hardener	1/4 kg

#### Table 1. Materials used for experiment



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V.RESULTS AND DISCUSSIONS

#### TENSILE TEST: LATHE OPERATION: Cutting speed S = $\pi$ DN/1000 = $\pi$ \*20\*300/1000 S = 9.424 m/minTime for Facing F = length of cut/ (feed\*rpm) = 9/(0.2\*300)F = 0.15 minBoth side of facing = 2\*0.15 = t1 = 0.3 min 4 no of facing = 4\*t1 = 4\*0.3T1 = 1.2minTime for Turning T = length of job to be turned / (feed\*rpm) =320/(0.2\*300)T = 5.333 min No of turning = 4\*T = 4\*5.333 = 50/(0.2\*300)4 k = 0.8333 Both side of knurling = $2^{k}$ = $2^{k}$ 0.833 K = 1.666 4 No of knurling = 4\*K = 4\*1.666T3 = 6.664 minMachining time T = T1+T2+T3=1.2+21.332+6.668 T = 29.2012 min**Tensile** Test 100% bounded of Steel vs. epoxy: (1) Area = { $\pi/4$ [(D<sup>2</sup>)-(d<sup>2</sup>)]} = $\pi/4$ [(28<sup>2</sup>)-(10<sup>2</sup>)]} = 537.21 mm<sup>2</sup> (2) Stress = Load / Area = 35000 / 537.21 = 65.151N/mm<sup>2</sup> (3) Strain = change length / original length = 210/200 = 1.05(4) Yield stress $\sigma y$ = Yield point / Initial Area= (3250)/ 537.21 = 60.497N/mm<sup>2</sup>

UTM MACHINE OPERATION:

Table 1: Results for Tensile Test

														%	of	%	of
														elon	gatio	Redi	JCt10
														n	in	n in	Area
		LEN	GTH	DIAM	ETER									len	gth		
S.NO MATERIAL		OF		OF		YIE	LD	ULTIN	<b>IATE</b>	BREA	KING	INTE	RFAC		C		
			SPECIMEN		SPECIMEN		٨D	LO	AD	LO	AD	IN	١G				
		(n	nm											(%)		(%)	
			)	(m	m)	(kg	f)	(kg	gf)	(kg	gf)	(k	gf)				
		100						100									
		%	50%	100%	50%	100%	50%	%	50%	100%	50%	100%	50%	100%	50%	100%	50
	5.1	• • •		-	,			10	20	2.6		•					
1	Polyester	208	210	1	6	33	24	42	38	36	35	30	25	4	10	92	94
	-						•	~ ~			~ ~			_		0.1	0.1
2	Epoxy	210	223	7.5	7.5	32.5	30	35	33	33	32	32	32.5	5	11	91. 7	91
																<i>'</i>	

(6) Nominal breaking Stress ( $\sigma_{bn}$ ) = Breaking Load / Initial Load =33000/537.21 = 61.428N/mm (7)Neck Area =  $\pi/4 d^2 = \pi/4 * (7.5)^2 = 44.17$ mm

(8) Actual breaking Stress ( $\sigma$ bn) = Breaking Load / Neck Area = 33000 / 44.178 = 747.11N/mm

(9) % Elongation in Length = (Final gauge length – Initial gauge length) /Initial gauge = (210-200) / 200

= 0.05 (5) %



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(10) % Reduction in Area = (Initial Area - Neck Area) / Initial Area = (537.21-44.178) / 537.21 = 0.917 (91.777) %= (91.777) %

#### Other calculated calues are shown in table 1.



Fig 2. Stresses vs. debounding length Curve: (a) Steel vs. epoxy, (b) Steel vs. Polyester

#### **TORSION TEST:**

LATHE OPERATION for Torsion Test: Time for facing = length of cut/ (feed\*rpm) = 6/ (0.2\*300) f =2\* 0.1 min= 0.2 min 4 material for facing = 4\*0.2 T1 = 0.8 min Time for turning = length of job to be turned/ (feed\*rpm) = 200/ (0.2\*300) t = 3.333 min 4 material for turning = 4\*t =4\*3.333, T2 = 13.332 min

Total Machine Time T1+T2 = 0.8+13.333 = 14.333 min



Fig 2. workpiece in Twisting Machine Table 2: Test Results for Torsion test

S	MATERIAL	LENO O	GTH F		INTE	RFACE	Modulus Of Elasticity		SHEAR STRESS (t)				
1. n o		SPECIMEN (mm)		Torque (kgf)		Degree $(\Phi)$		Radian		(C) (Kpa)		(N/mm <sup>2</sup> )	
		100%	50%	100%	50%	100%	50%	100%	50%	100%	50 %	100%	50%
1	EPOXY	182	172	204	202	150	130	2.618	2.268	2.316	2.52	471.68	466.96
2	POLYESTER	178	176	239	199	178	90	3.108	1.570	3.684	2.26	460.126	552.57

#### **MODEL** Calculation for Twisting operation:

100% bounded: (steel vs. epoxy)

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Shear Stress ( $\tau$ ): T=  $\pi/16^*\tau [D^4) \cdot (d^4)/D$ ] = (2.00\*10^6) =  $\pi/16^*\tau [((28^4) \cdot (10^4))/28]$   $\tau = 471.68$ Nmm<sup>2</sup> Polar Moment of Inertia: J =  $\pi/32 [(D^4) \cdot (d^4)] = \pi/32 [(28^4) \cdot (10^4)] = (59.36^*10^4) mm4$ Modulus of Elasticity: T/J = C\* $\theta$  /L (2.00\*10^6)/(59.36\*10^3) = [(C\*2.618)/180] C = 33.692\*180/2.618 = 6064.690/2.618t C = 2.31653kpa 50% bounded: (steel vs. epoxy) Shear stress ( $\tau$ ): T =  $\pi/16^*\tau [D^4) \cdot (d^4)/D$ ] (1.98\*10^6) =  $\pi/16^*\tau [((28^4) \cdot (10^4))/28]$   $\tau = 466.96$ Nmm<sup>2</sup> Polar Moment of Inertia: J =  $\pi/32 [(D^4) \cdot (d^4)] = -\pi/32 [(28^4) \cdot (10^4)]$  J = (59.36\*10^3) mm4 Modulus of Elasticity: T/J = C\* $\theta/L$ (1.98\*10^6) / (59.36\*10^3) = [(C\*2.268) / 172]

C = 33.355\*172/2.268 = 2.52956kpa

#### Other calculated values are shown in table 2.

#### **IMPACT TEST:**

Impact test is done by using Izoid and Charpy Impact test experiments.

Model calculation for Impact Test:

100% bounded: (steel vs. polyester)

V = 1\*b\*d = 56\*8\*8.5 = 38,080 mm3

Impact Modulus = 90 J

#### Other calculated values for Epoxy, Dip on Polyester, and Normal MS Materials are tabulated in table 3. Table 3: Impact Test Results

S NO	MATERIAL	TESTING	LENGTH (mm)	BREATH (mm)	DEAPTH (mm)	VOLUME (mm^3)	IMPACT Strength (J)
1	POLYESTER	IZOD	56	8.4	8	3673.6	90
2	DIP ON						
	POLYESTER	IZOD	56	8.2	8	3763.2	88
3	EPOXY	CHARPY	56	8.5	8	3808	112
4	NORMAL MS	CHARPY	56	8.0	8	3584	110

#### **VI.CONCLUSION**

The Application of composite materials in modern world isincrease now, for example aircrafts windows frames, racing car, etc., If a 20% of debonding in the composite material will increase the stress inside the composite in our experimental setup we tested in composite material more than 20% debonding which resulted in immediate breakdown in the material and the stress created inside the material is high.



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